

<b>APJ ABDUL KALAM TECHNOLOGICAL UNIVERSITY</b>		
THIRD SEMESTER B.TECH DEGREE(S) EXAMINATION, MAY2019		
<b>Course Code: ME203</b>		
<b>Course Name: MECHANICS OF FLUIDS</b>		
Max. Marks: 100		Duration: 3 Hours
<b>PART A</b>		
<i>Answer any three full questions, each carries 10marks.</i>		

1. (a) statement 2 marks, Proof 3 marks

$$(b) \tau = \mu (du/dy) = 1591.6 \text{ N/m}^2 \quad F = \tau \times A = 180 \text{ NT} = F \times D/2 = 36 \text{ Nm}$$

$$\text{Power lost} = T \times \omega = T (2\pi N/60) = 716.283 \text{ W}$$

6 marks

2. Derivation - Total pressure 5 mark

Derivation - Centre of pressure 5 mark

3. Classification of fluid flow 10 marks

4. (a) Velocity vector for 2D is given by

$$V = u\mathbf{i} + v\mathbf{j}$$

Case a :

The continuity equation for steady, 2D incompressible flow is given by

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} = 0$$

The continuity equation for steady incompressible flow is satisfied hence the flow is possible

5 mark

Case b:

The expression for rotation of a fluid particle in the X-Y plane about Z axis is given by

$$w_z = \frac{1}{2} \left( \frac{\partial v}{\partial x} - \frac{\partial u}{\partial y} \right)$$

$$w_z = 0$$

hence the flow is irrotational.

5 mark

<b>PART B</b>
<i>Answer any three full questions, each carries 10marks.</i>

5. Hagen Poiseuille equation Derivation 10 mark

6. Bernoulli's equation 2 marks

$$Q = a_1 v_1 = a_2 v_2 = 5400 \text{ L/min} = 0.09 \text{ m}^3/\text{s}$$

$$V_1 = 0.079 \text{ m/s}, V_2 = 0.319 \text{ m/s}, \text{ Slope} = 1 \text{ in } 100$$

$$\tan\theta = 1/100 = 0.01 \quad \theta = 0.572^\circ$$

$$Z_1 = 3\text{m}, Z_2 = 0$$

$$P_1/\rho g + V_1^2/2g + Z_1 = P_2/\rho g + V_2^2/2g + Z_2$$

$$P_2 = 98.046 \text{ kPa}$$

Pressure at the lower end. 8 mark

7.  $V_{\max} = 2 \text{ m/s}$  Average velocity,  $V = V_{\max}/2 = 1 \text{ m/s}$   
 $Re = \rho V d/\mu$  Flow is Laminar  
Pressure gradient in the direction of flow,  $\Delta P/L = 32\mu V/d^2 \text{ N/m}^2$  per m length of pipe  
shear stress at the pipe wall,  $\tau = (\Delta P/L) r/2 \text{ N/m}^2$   
velocity at a distance 30 mm from the wall =  $(1/8\mu) (\Delta P/L)r^2 \text{ m/s}$
- Average velocity 2 marks  
(i) the pressure gradient in the direction of flow (ii) shear stress at the pipe wall; (iii) Reynold's number; and (iv) velocity at a distance 30 mm from the wall. (4 x 2 = 8 marks),
8. Sketch of venturimeter 2 marks  
Bernoulli's equation 2 marks  
Rate of flow - Derivation 6 marks

### PART C

*Answer any four full questions, each carries 10marks.*

9. boundary layer 2marks  
Sketch 3 marks  
Illustration 5 marks
10. Sketch – 2 marks,  
Boundary Layer Thickness  
Laminar Boundary Layer  
Turbulent boundary Layer  
Laminar Sub Layer (2x4 = 8 marks)
11.  $Re = VL/\nu = 400000$ , laminar 1 mark  
Blasius relations 1 mark each (1x3 = 3 marks)  
(a) Boundary layer thickness at the end of the plate  
 $\delta = 5x/\sqrt{Re} = 7.906 \times 10^{-3} \text{ m}$   
(b) Shear stress at the middle of the plate  
 $\tau = 0.332\rho U^2/\sqrt{Rex} = 0.0321 \text{ N/m}^2, x = 0.5 \text{ m}$   
(c) Total drag per unit length on the sides of the plate,  
 $C_D = 1.328/\sqrt{Re} = 2.0997 \times 10^{-3}$

$$F_D = (1/2 \rho A U^2) \times C_D = 0.04535 \text{ N}$$

Answer (3 x 2 = 6 marks)

12. Buckingham's  $\pi$  theorem 3 marks, Relation 7 marks

$$\Delta P = f(D, L, \rho, \mu, V, K)$$

$$n = 7, m = 3, n - m = 7 - 3 = 4$$

$$f(\pi_1, \pi_2, \pi_3, \pi_4) = 0,$$

$$\pi_1 = \Delta P / \rho V^2, \pi_2 = \frac{L}{D}, \pi_3 = \mu / \rho V D, \pi_4 = K/D$$

13. Hydraulic similarities 3 types 3 marks

Explanations 7 marks

14. Model law- 3 marks

$$L_p/L_m = 40/2.5$$

Froude's model law

$$(F_e)_{\text{model}} = (F_e)_{\text{prototype}}$$

$$[V/\sqrt{gL}]_{\text{model}} = [V/\sqrt{gL}]_{\text{prototype}}, L_p/L_m$$

$$V_p/V_m = \sqrt{\left(\frac{L_p}{L_m}\right)}$$

velocity of prototype,  $V_p$  3 marks

$$\rho_p/\rho_m = 1025/1000$$

$$F_p/F_m = (\rho L^2 V^2)_{\text{prototype}} / (\rho L^2 V^2)_{\text{model}}$$

Force required to drive the prototype, Force - 4 marks